



Geometrically nonlinear finite element modelling of linear elastic truss structures

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1.5. Geometrically nonlinear bar finite element

Inspired and adapted from the 'Nonlinear Modeling of Structures' course of Prof. Thierry J. Massart at the ULB



Derivation of F_{int} and K_{el}

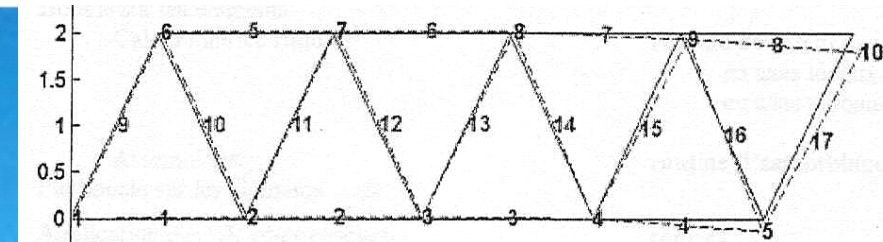
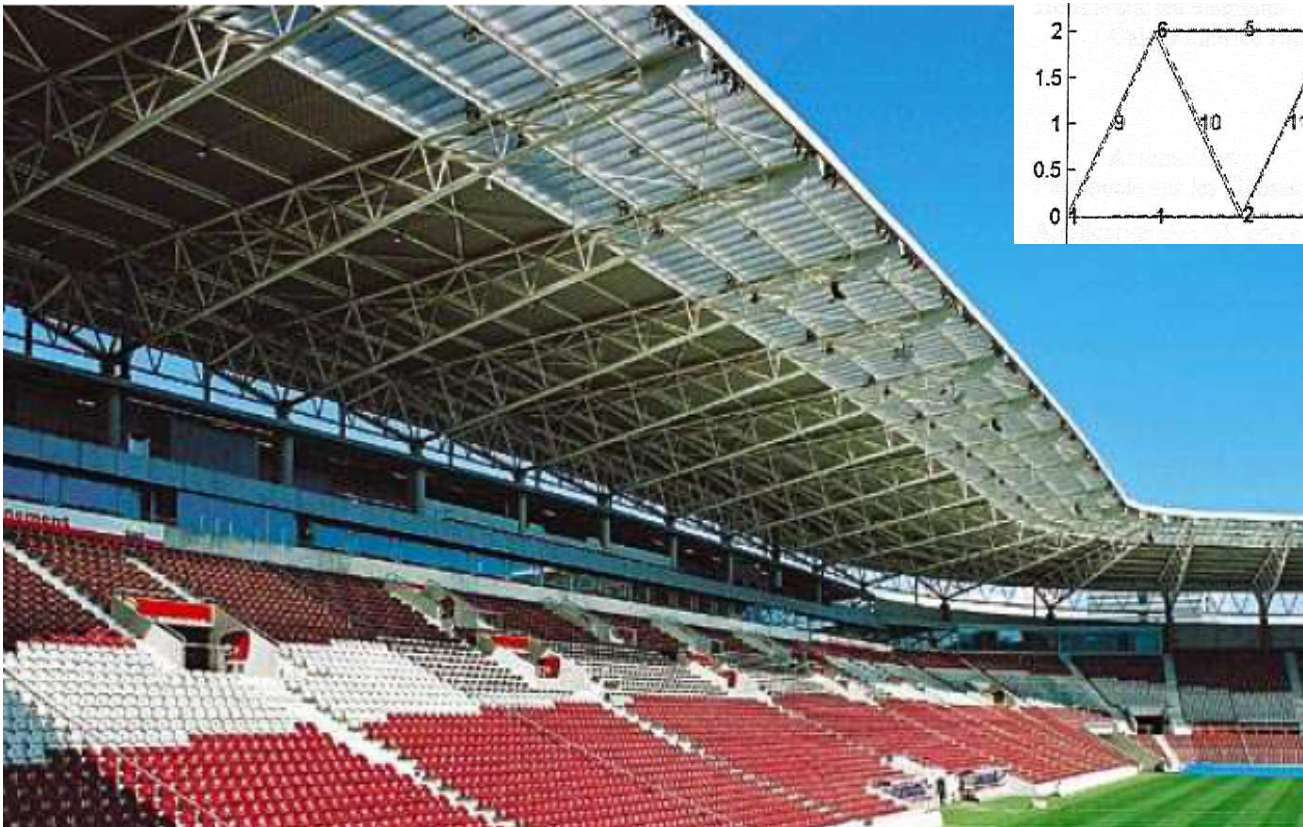
Lab: Complete missing element relationships

Analyse the 'V' shaped structure

Problem statement

Determine the displacements of structures at equilibrium

Large displacements and small deformations, linear elastic material



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Nonlinear response?

Linear elastic computation



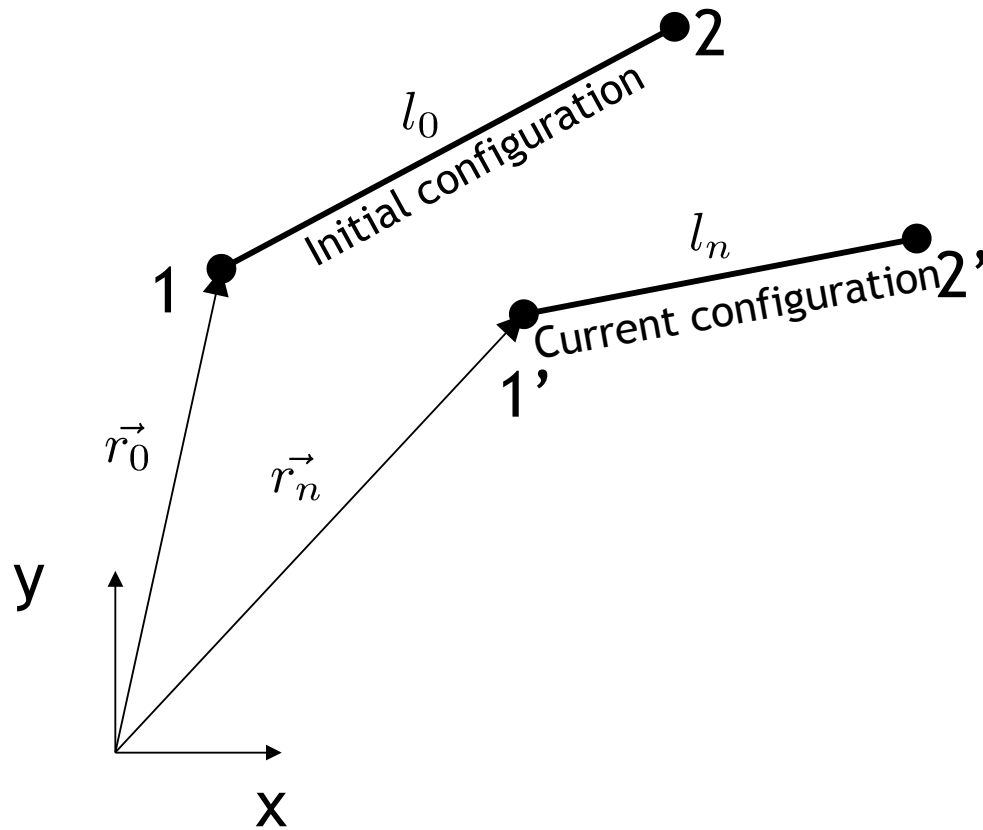
Stadium of Geneva



Kuala Terengganu stadium, Malaysia, 2009



Nonlinear response



Kinematics of the bar

$$\vec{r}^T = \{x, y\} \quad \vec{u}^T = \{u, v\}$$

$$\vec{r}_n = \vec{r}_0 + \vec{u}$$

$$\mathbf{x}^T = \{x_1, x_2, y_1, y_2\} \text{ nodal coordinates}$$

$$\mathbf{u}^T = \underbrace{\{u_1, u_2, v_1, v_2\}}_{\text{nodal displacements}}$$

Order of dof!

Defining

$$\mathbf{x}_{21}^T = \{(x_2 - x_1), (y_2 - y_1)\}$$

$$\mathbf{u}_{21}^T = \{(u_2 - u_1), (v_2 - v_1)\}$$

Initial and current length of the bar

$$l_0^2 = 4 \alpha_0^2 = \mathbf{x}_{21}^T \mathbf{x}_{21}$$

$$l_n^2 = 4 \alpha_n^2 = (\mathbf{x}_{21} + \mathbf{u}_{21})^T (\mathbf{x}_{21} + \mathbf{u}_{21})$$

‘Green’ strain definition

$$\epsilon_G = \frac{l_n - l_0}{l_0} = \frac{(l_n - l_0)(l_n + l_0)}{l_0 (l_n + l_0)} = \frac{l_n^2 - l_0^2}{l_0^2 (2 + \epsilon_G)} \quad \text{with } \epsilon_G \text{ small}$$

$$\epsilon_G = \frac{l_n^2 - l_0^2}{2l_0^2} = \frac{(\mathbf{x}_{21} + \mathbf{u}_{21})^T (\mathbf{x}_{21} + \mathbf{u}_{21}) - \mathbf{x}_{21}^T \mathbf{x}_{21}}{2 \mathbf{x}_{21}^T \mathbf{x}_{21}}$$

$$\text{because } \begin{cases} l_0^2 = 4 \alpha_0^2 = \mathbf{x}_{21}^T \mathbf{x}_{21} \\ l_n^2 = 4 \alpha_n^2 = (\mathbf{x}_{21} + \mathbf{u}_{21})^T (\mathbf{x}_{21} + \mathbf{u}_{21}) \end{cases}$$

with

$$\mathbf{x}_{21}^T = \{(x_2 - x_1), (y_2 - y_1)\}$$

$$\mathbf{u}_{21}^T = \{(u_2 - u_1), (v_2 - v_1)\}$$

Link between $\delta\epsilon$ - $\delta\mathbf{u}$ and σ - ϵ

$$\epsilon_G = \frac{l_n^2 - l_0^2}{2l_0^2} = \frac{(\mathbf{x}_{21} + \mathbf{u}_{21})^T (\mathbf{x}_{21} + \mathbf{u}_{21}) - \mathbf{x}_{21}^T \mathbf{x}_{21}}{2 \mathbf{x}_{21}^T \mathbf{x}_{21}}$$

$$\mathbf{b}_1^T = \frac{1}{4\alpha_0^2} \{-x_{21}, x_{21}, -y_{21}, y_{21}\}$$

$$\text{with } \begin{cases} x_{21} = x_2 - x_1 \\ y_{21} = y_2 - y_1 \\ \dots \end{cases}$$

$$\mathbf{b}_2(\mathbf{u})^T = \frac{1}{4\alpha_0^2} \{-u_{21}, u_{21}, -v_{21}, v_{21}\}$$

Link between the variation of displacements and the variation of strain

$$\delta\epsilon_G = \frac{\partial\epsilon_G}{\partial\mathbf{u}} \delta\mathbf{u} = (\mathbf{b}_1 + \mathbf{b}_2(\mathbf{u}))^T \delta\mathbf{u} = \mathbf{b}^T \delta\mathbf{u}$$

Link between stress and strain

$$\sigma_G = \epsilon_G E \quad \text{since linear elastic material}$$

Expression of the internal force vector

Virtual work theorem $\delta \mathbf{u}_v$ is the virtual displacement vector

$$\sum_e \delta \mathbf{u}_v^T \mathbf{f}_{int} = \sum_e \int \sigma_G \delta \epsilon_v dV_0 = \sum_e \delta \mathbf{u}_v^T \int \sigma_G \mathbf{b} dV_0$$

because

$$\delta \epsilon_G = \frac{\partial \epsilon_G}{\partial \mathbf{u}} \delta \mathbf{u} = (\mathbf{b}_1 + \mathbf{b}_2(\mathbf{u}))^T \delta \mathbf{u} = \mathbf{b}^T \delta \mathbf{u}$$

$$\sigma_G = \epsilon_G E$$

Expression of the internal forces

$$\mathbf{f}_{int} = \int \sigma_G \mathbf{b} dV_0 = 2\alpha_0 A_0 \sigma_G \mathbf{b}$$

element cross-section

Derivation of the stiffness matrix (I/II)

Internal force vector

$$\mathbf{f}_{int} = \int \sigma_G \mathbf{b} dV_0 = 2\alpha_0 A_0 \mathbf{b} \sigma_G$$

Stiffness matrix of a bar

$$\mathbf{K}_t = \frac{\partial \mathbf{f}_{int}}{\partial \mathbf{u}} = 2\alpha_0 A_0 \mathbf{b} \frac{\partial \sigma_G}{\partial \mathbf{u}} + 2\alpha_0 A_0 \frac{\partial \mathbf{b}}{\partial \mathbf{u}} \sigma_G$$

$$\mathbf{K}_t = \mathbf{K}_t^s + \mathbf{K}_t^g$$

Contribution of the stress change in a bar

$$\frac{\partial \sigma_G}{\partial \mathbf{u}} = E \frac{\partial \epsilon_G}{\partial \mathbf{u}} = E \mathbf{b}(\mathbf{u})^T$$

$$\mathbf{K}_t^s = 2\alpha_0 A_0 \mathbf{b} \frac{\partial \sigma_G}{\partial \mathbf{u}} = 2\alpha_0 A_0 E \mathbf{b} \mathbf{b}^T$$

Derivation of the stiffness matrix (II/II)

Contribution of the change in the geometry

$$\frac{\partial \mathbf{b}}{\partial \mathbf{u}} = \frac{\partial \mathbf{b}_2}{\partial \mathbf{u}} = \frac{1}{4\alpha_0^2} \mathbf{S} \quad \mathbf{b}_2(\mathbf{u})^T = \frac{1}{4\alpha_0^2} (-u_{21}, u_{21}, -v_{21}, v_{21})$$

$$\mathbf{S} = \begin{bmatrix} 1 & -1 & 0 & 0 \\ -1 & 1 & 0 & 0 \\ 0 & 0 & 1 & -1 \\ 0 & 0 & -1 & 1 \end{bmatrix}$$

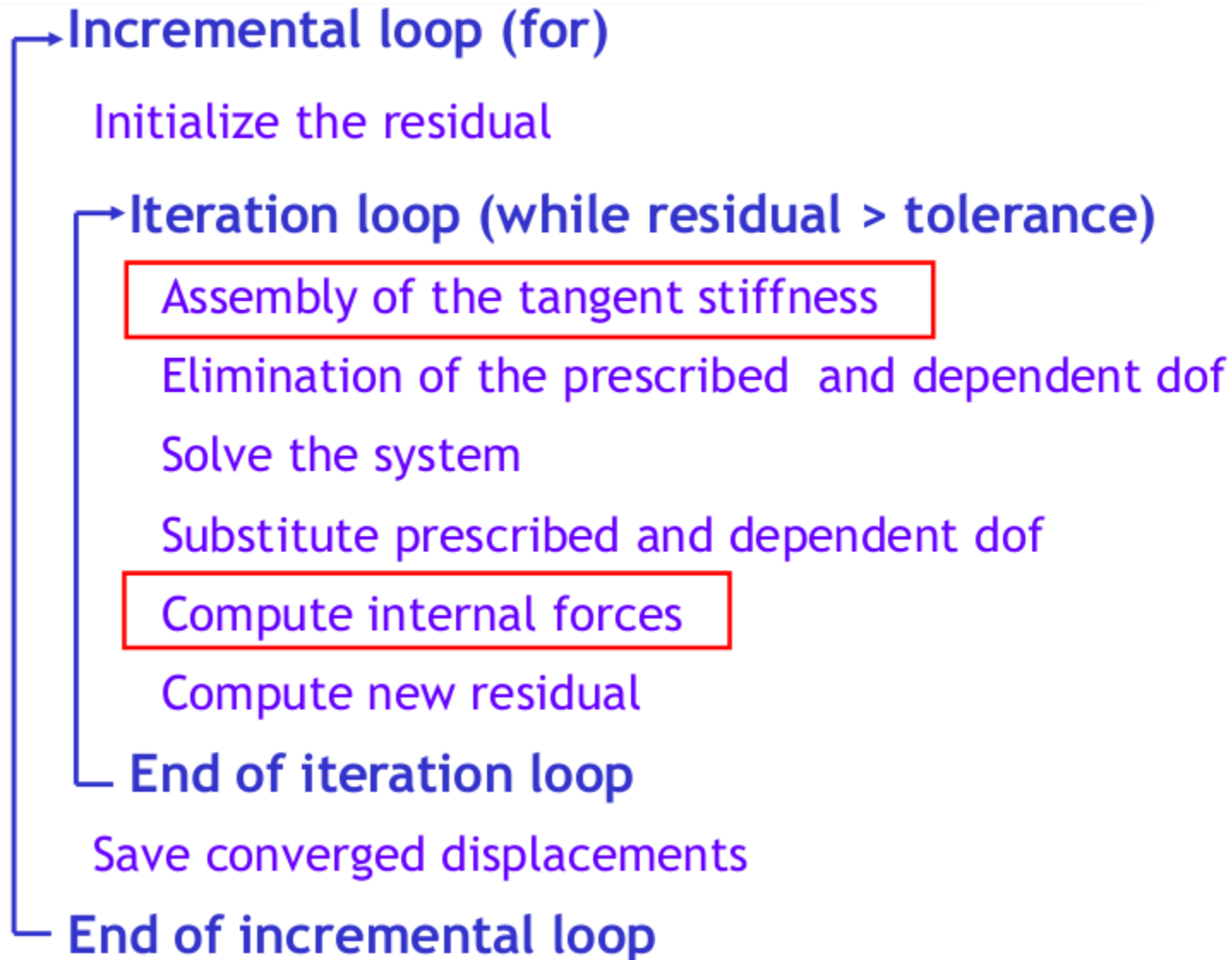
$$\mathbf{K}_t^g = 2\alpha_0 A_0 \frac{\partial \mathbf{b}}{\partial \mathbf{u}} \sigma_G = \frac{A_0 \sigma_G}{2\alpha_0} \mathbf{S}$$

Expression of the stiffness matrix of a bar

$$\mathbf{K}_t = 2\alpha_0 A_0 E \mathbf{b} \mathbf{b}^T + \frac{A_0 \sigma_G}{2\alpha_0} \mathbf{S}$$



Use of the element in the NL code



NL structural response

